

# Global Attenuation of Acoustic Fields Using Energy-Based Active Control Techniques

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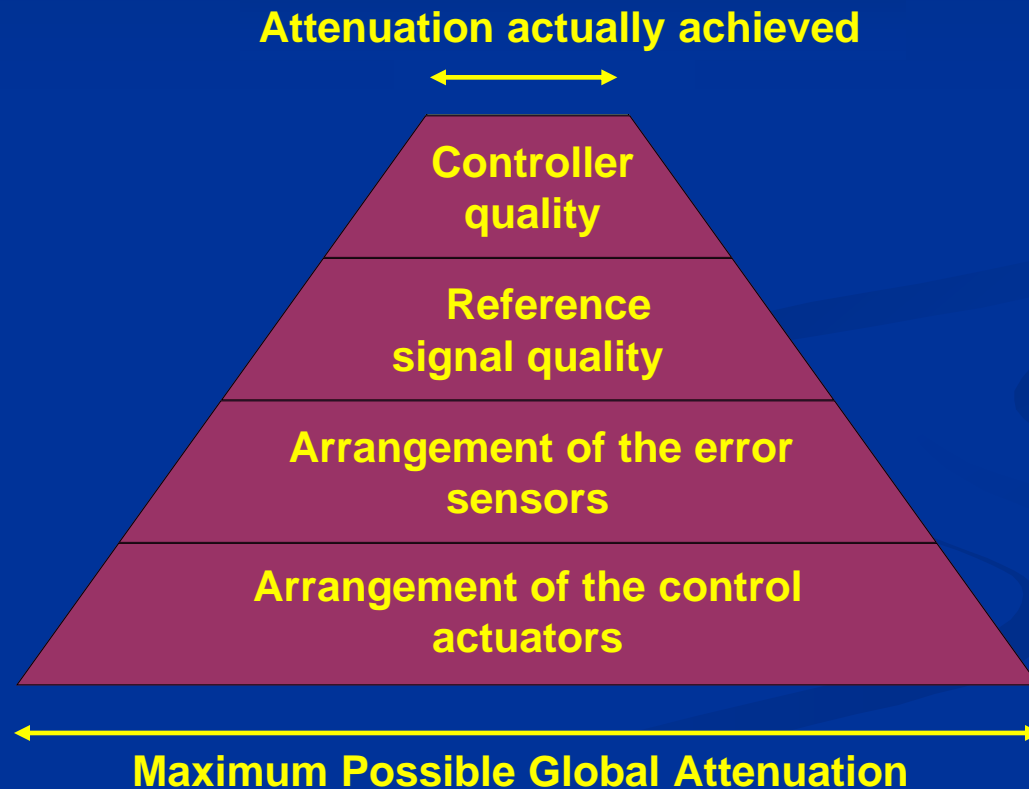
# Outline

- Local vs. global control – identifying ANC objective
- Global control concepts
  - Free field radiation
  - Enclosed sound fields
- Implementation of global control
  - Free field radiation
  - Enclosed sound fields
- Application results
  - Global control of axial fan noise
  - Global control of cab noise in heavy equipment

# Local vs. Global Control

- Exactly what are we trying to accomplish?
  - Does local control meet my objective?
  - Do I need a global solution?
- If objective not kept in mind, probability is high that one will not accomplish desired outcome
- If global control desired for application, energy in the field must be minimized
  - Implications of how this is done vary by application

# Hierarchy of Active Control (Snyder, Hansen)



# Global Control – Free Field Radiation

- Physical mechanism for global control is source coupling
  - Function of frequency and source spacing/location
- Need to minimize overall power (energy) radiated into field
- Concept easily seen for case of two monopole sources

$$\Pi = \frac{\rho c k^2}{8\pi} |Q_1|^2 \left[ 1 + A^2 + 2A \frac{\sin(kd)}{kd} \cos \gamma \right]$$

$$\Pi_{\min} = \frac{\rho c k^2}{8\pi} |Q_1|^2 \left[ 1 - \left( \frac{\sin(kd)}{kd} \right)^2 \right]$$

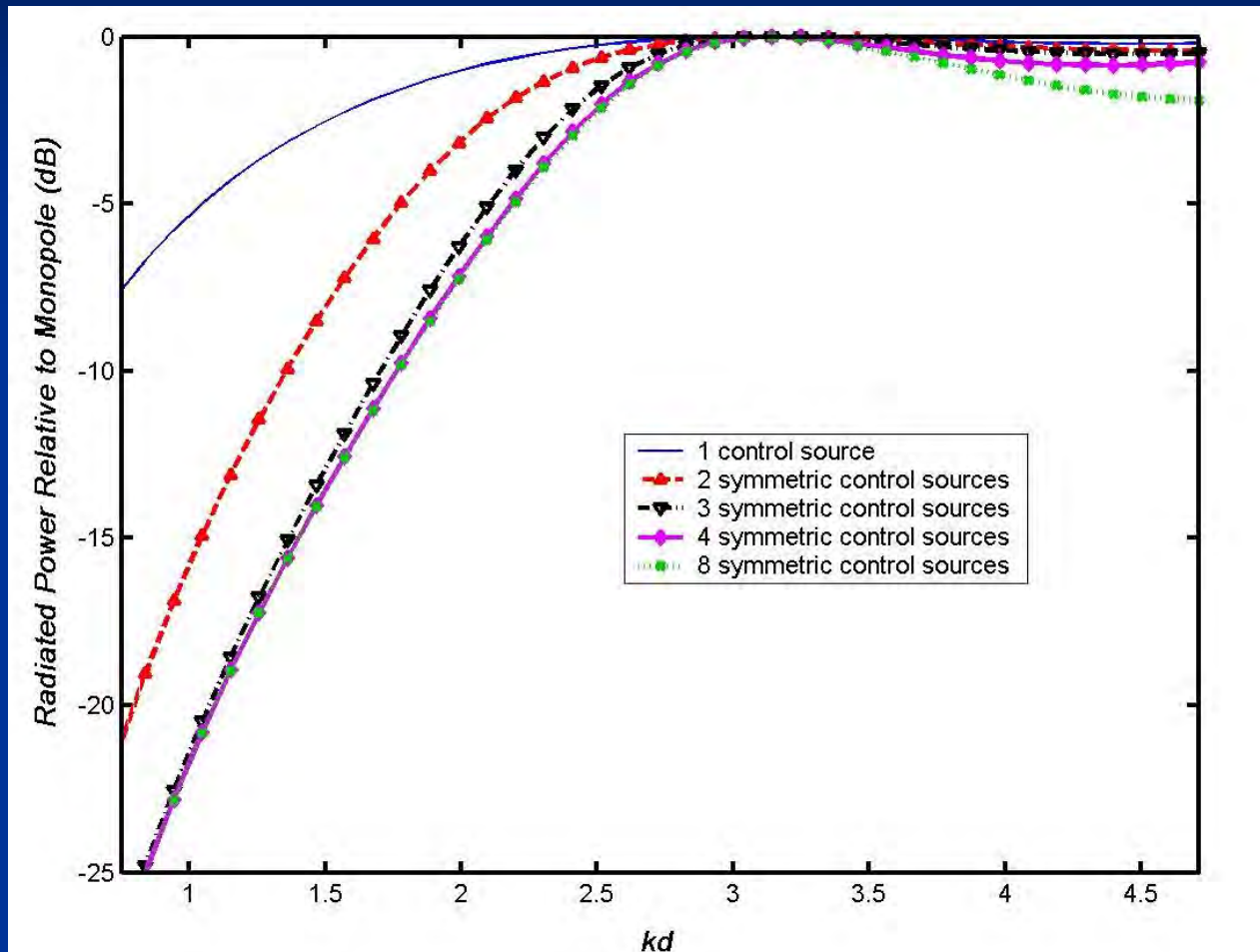
- For additional and/or more complex sources, same concept holds, but results expressed in terms of matrices



# Free Field Radiation – cont.

- Control that can be achieved is a function of
  - Source spacing
  - Frequency
  - Number of control sources used
  - Geometric configuration of sources
- Power attenuation can be determined for various configurations
- Optimization is a more difficult problem, but needs to be looked at for maximum attenuation

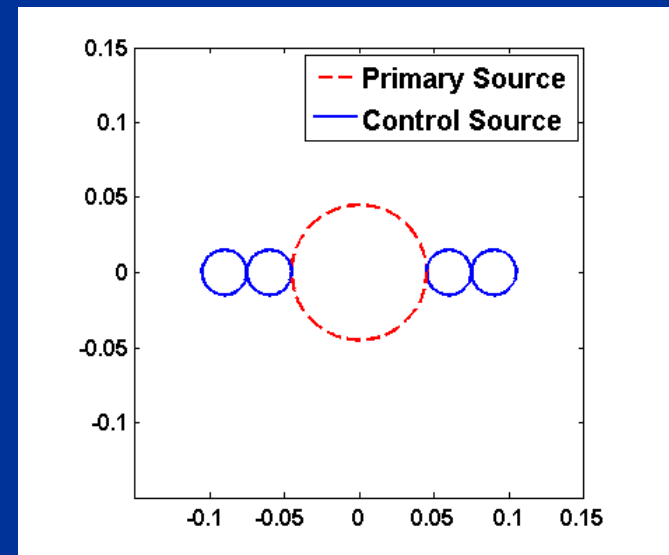
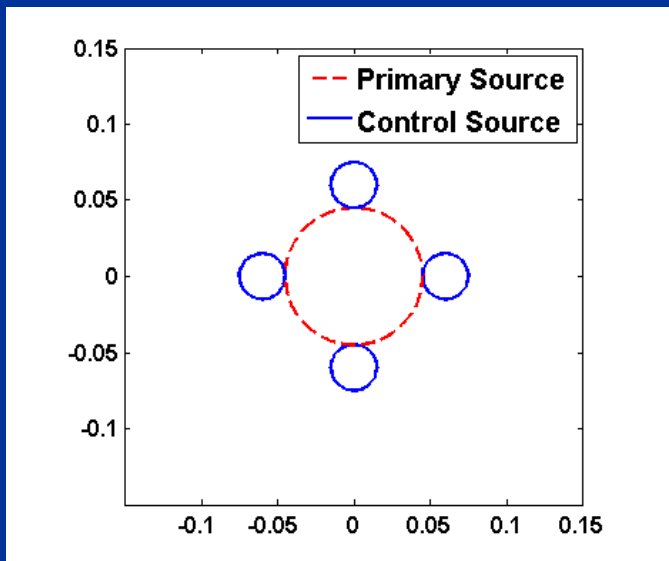
# Free Field Radiation – cont.



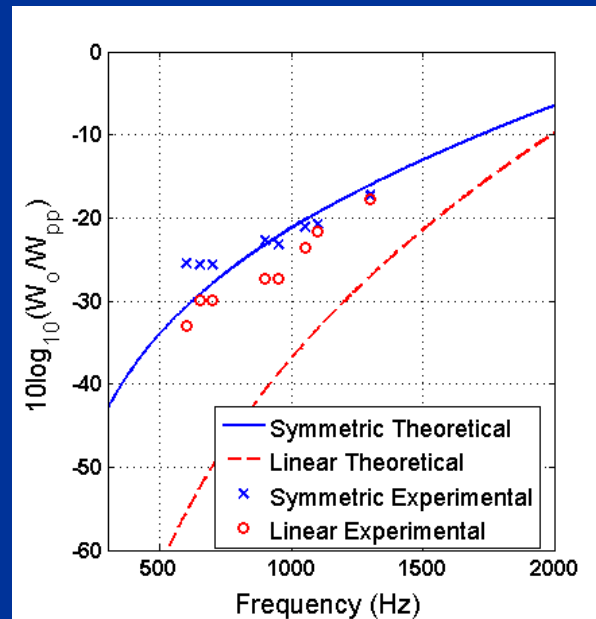
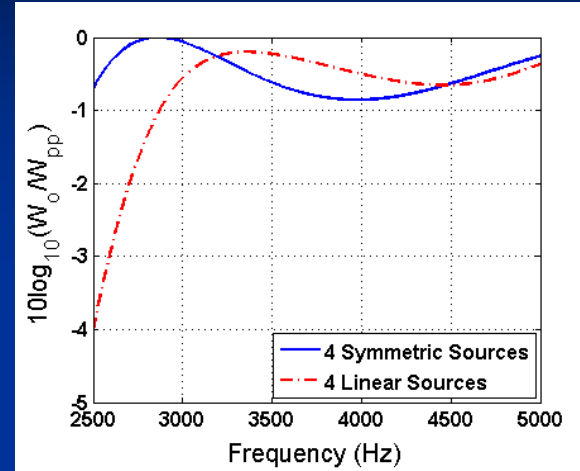
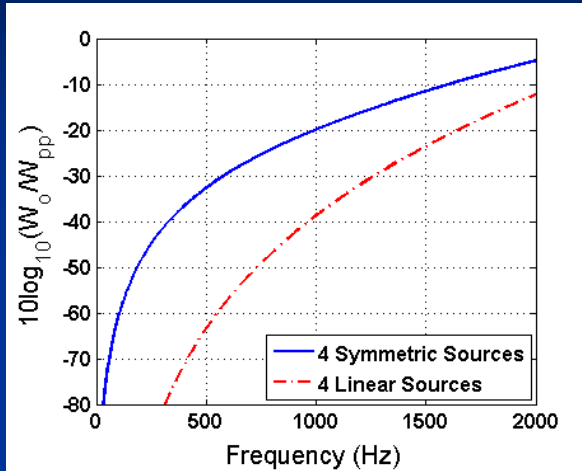
Power attenuation achieved for control sources in the plane of primary source

# Optimal Control Source Configuration

- Genetic algorithm used – look at potential source configurations, subject to constraints



# Comparisons of Symmetric/Linear Arrays



# Global Control – Enclosed Sound Fields

- Global control still occurs through mechanism of source coupling, but in this case, sources couple through modes of enclosure
- Suggested that optimal solution for global control is to minimize overall potential energy (Nelson and Elliott)
- Requires integration of global energy, which is not practical – but starting point to determine system configuration

# Enclosed Sound Fields – cont.

- Formally, can represent solution in terms of modes (single primary source, single secondary source)

$$\hat{p}(\mathbf{x}) = \sum_{N=1}^{\infty} \hat{A}_N + \hat{B}_N \hat{Q}_c \hat{\Psi}_N(\mathbf{x})$$

$$\hat{Q}_c = -\frac{\sum_{N=1}^{\infty} \hat{A}_N \hat{B}_N^* \Lambda_N}{\sum_{N=1}^{\infty} \hat{B}_N \hat{B}_N^* \Lambda_N}$$

- Depends on all modal amplitudes, but does not have any spatial dependence
- For additional sources, matrix formulation applies

# Enclosed Sound Fields – cont.

- Control solution depends on  $\hat{A}_N, \hat{B}_N$

$$\hat{A}_N = j\rho ck Q_p(\omega) \frac{\Psi_N(\mathbf{x}_p)}{V\Lambda_N(k_N^2 - k^2)}$$

$$\hat{B}_N = j\rho ck \frac{\Psi_N(\mathbf{x}_c)}{V\Lambda_N(k_N^2 - k^2)}$$

- Dependence on
  - Frequency
  - Source location(s), relative to primary source location(s)
  - Number of sources

# Error Sensor Design

- Previous results focused on source configuration – first step for achieving global control
- For implementation, requires measurement of sound power (free field radiation) or global potential energy (enclosed fields)
- These quantities not available in practice. Need to develop sensing schemes that approximate desired quantity as best we can
- Appropriate sensing scheme depends on application
- Example: Area (and shaped) piezofilm sensors for ASAC (Berry, Nelson & Elliott, Sommerfeldt); radiation mode sensing (Nelson & Elliott, Hansen, others)

# Sensor Implementation – Free Field Radiation

- Application of interest is radiation from small axial fans
- Sound power is not easily available, particularly in real time
- Far field microphones could be used – not practical if a compact solution is desired
- Sensor approach developed based on microphones in near field

# Sensor Implementation – Free Field Radiation

- Approach based on analysis of power minimization
- After source configuration determined, analytical/numerical model of system developed and power radiation calculated
- Power minimization carried out to determine optimal source strength(s). For single control source,

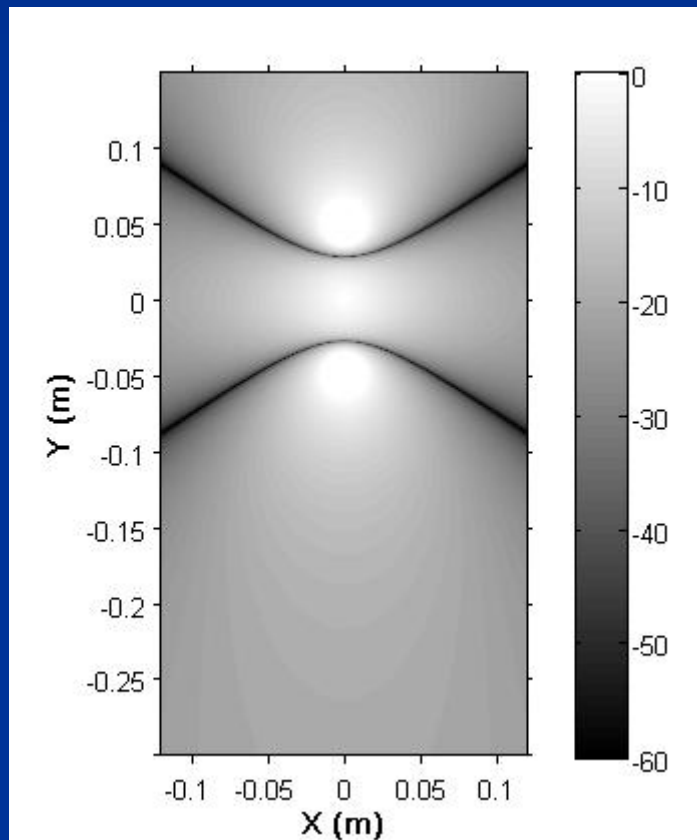
$$Q_2 = -Q_1 \frac{\sin(kd)}{kd}$$

# Sensor Implementation – Free Field Radiation

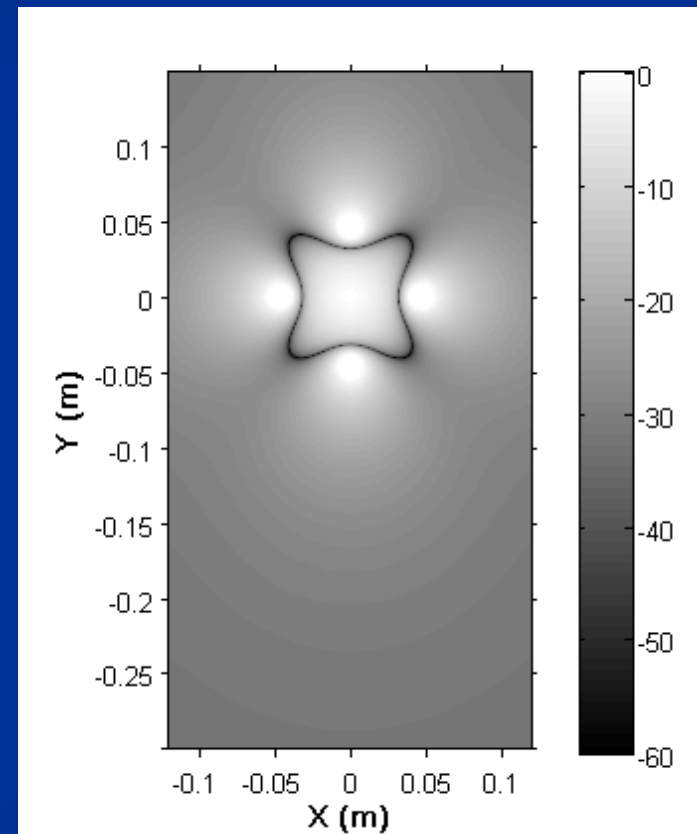
- Using optimal source strength(s), near field pressure can be determined.
- Optimal error sensor locations are those where pressure attenuation is greatest when implementing optimal source strength(s).
- Near field nulls can be identified to indicate suitable sensor locations

# Sensor Implementation – Free Field Radiation

Pressure Field Reduction With Optimal Control Implementation  
(600 Hz,  $d = 0.045\text{m}$ )



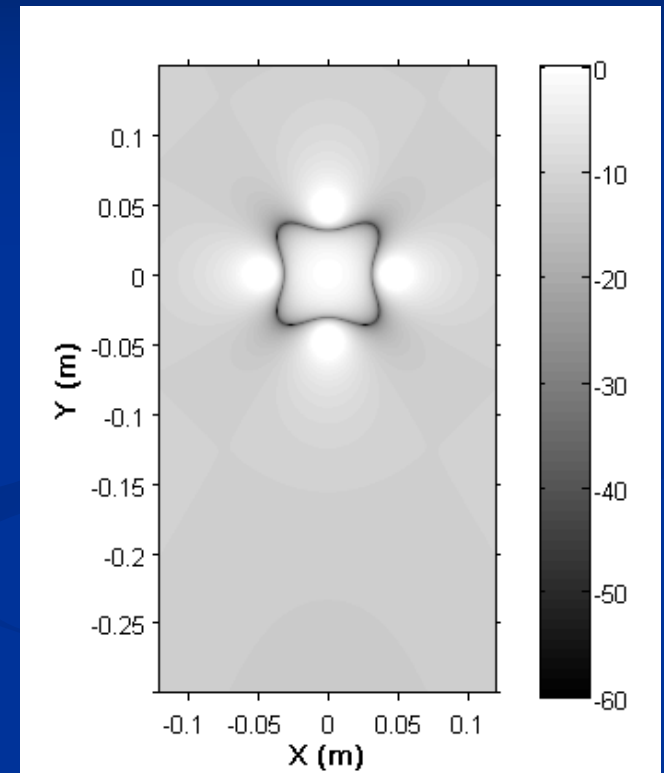
2 Control Sources



4 Control Sources

# Sensor Implementation – Free Field Radiation

- For co-planar sources, has been shown that nulls in plane of sources represent optimal locations (deepest nulls)
- Nulls have weak frequency dependence
- Method does rely on reasonable model of source radiation characteristics



1800 Hz

# Sensor Implementation – Enclosed Sound Field

- Ideally, need to measure global potential energy
  - Implementations with this strategy often result in high channel count for error sensors
- Alternative is to seek local measurement that provides information that is more global in nature
- One such possibility is the use of energy density
  - Sommerfeldt, Nashif, Parkins, Faber (Penn State/BYU)
  - Hansen, Cazzolato (University of Adelaide)

$$e = \frac{p}{2\rho c^2} + \frac{\rho}{2} \hat{v}^2$$

# Sensor Implementation – Enclosed Sound Field

- Energy density for  $N$ th mode given by

$$\begin{aligned} e_N(\mathbf{x}) &= \frac{\hat{p}_N^2(\mathbf{x})}{2\rho c^2} + \frac{\rho}{2} \hat{v}^2(\mathbf{x}) \\ &= \frac{A_N^2}{2\rho c^2} \frac{1}{k^2} \left( k_x^2 \cos^2(k_y y) \cos^2(k_z z) + k_y^2 \cos^2(k_x x) \cos^2(k_z z) + k_z^2 \cos^2(k_x x) \cos^2(k_y y) \right) \end{aligned}$$

- Velocity estimated using pressure gradient

$$\begin{aligned} v_x &= -\frac{1}{j\omega\rho} \frac{\partial p}{\partial x} \\ &\approx -\frac{1}{j\omega\rho} \frac{p_2 - p_1}{\Delta x} \end{aligned}$$

# Sensor Implementation – Enclosed Sound Field

- Prototype ED sensors



University of  
Adelaide



BYU – 2D



BYU – 3D

# Sensor Implementation – Enclosed Sound Field

- What makes ED sensors potentially better?
- Less spatial dependence for ED
- For 1D field (or axial modes), ED is uniform for mode
- For active control, need to have sensors away from nodes in order to effectively attenuate that mode

# Control Solutions for Possible Control Strategies

- Global Potential Energy

$$\hat{Q}_c = -\frac{\sum_{N=1}^{\infty} \hat{A}_N \hat{B}_N^* \Lambda_N}{\sum_{N=1}^{\infty} \hat{B}_N \hat{B}_N^* \Lambda_N}$$

- Squared Pressure

$$\hat{Q}_c = -\frac{\sum_{N=1}^{\infty} \hat{A}_N \Psi_N(\mathbf{r}_e)}{\sum_{N=1}^{\infty} \hat{B}_N \Psi_N(\mathbf{r}_e)}$$

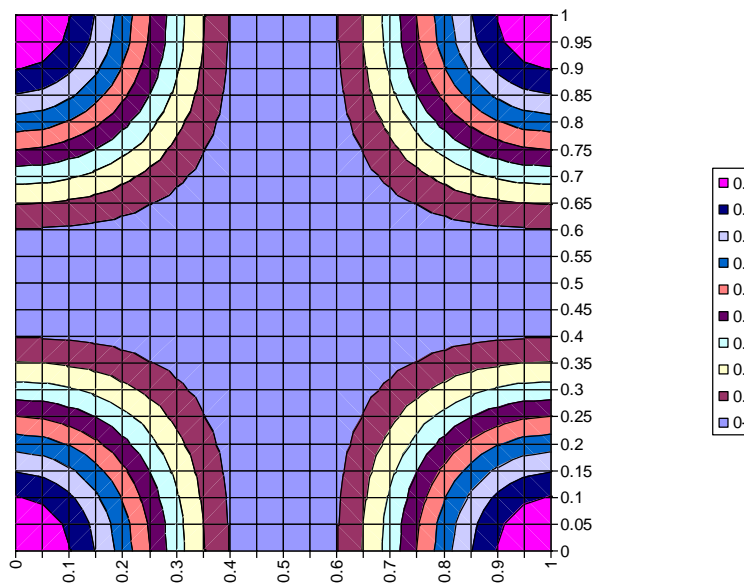
- Energy Density

$$\hat{Q}_c = -\frac{\sum_{N=1}^{\infty} \sum_{L=1}^{\infty} \hat{A}_N \hat{B}_L^* F_{NL}}{\sum_{N=1}^{\infty} \sum_{L=1}^{\infty} \hat{B}_N \hat{B}_L^* F_{NL}}$$

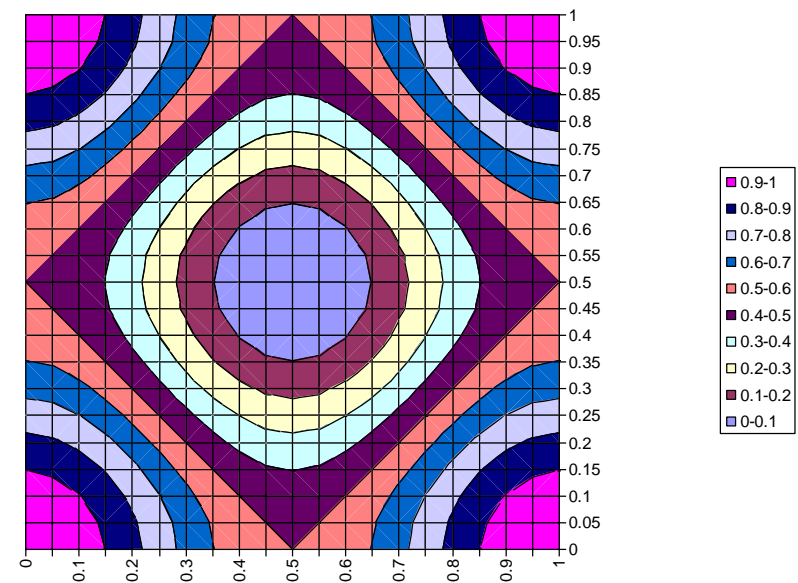
$$F_{NL} = \Psi_N(\mathbf{r}_e) \Psi_L(\mathbf{r}_e) + \frac{1}{k^2} \left( \frac{\partial \Psi_N(\mathbf{r}_e)}{\partial x} \frac{\partial \Psi_L(\mathbf{r}_e)}{\partial x} + \frac{\partial \Psi_N(\mathbf{r}_e)}{\partial y} \frac{\partial \Psi_L(\mathbf{r}_e)}{\partial y} + \frac{\partial \Psi_N(\mathbf{r}_e)}{\partial z} \frac{\partial \Psi_L(\mathbf{r}_e)}{\partial z} \right)$$

# Spatial Dependence of Pressure vs. Energy Density

(1,1,0) Mode

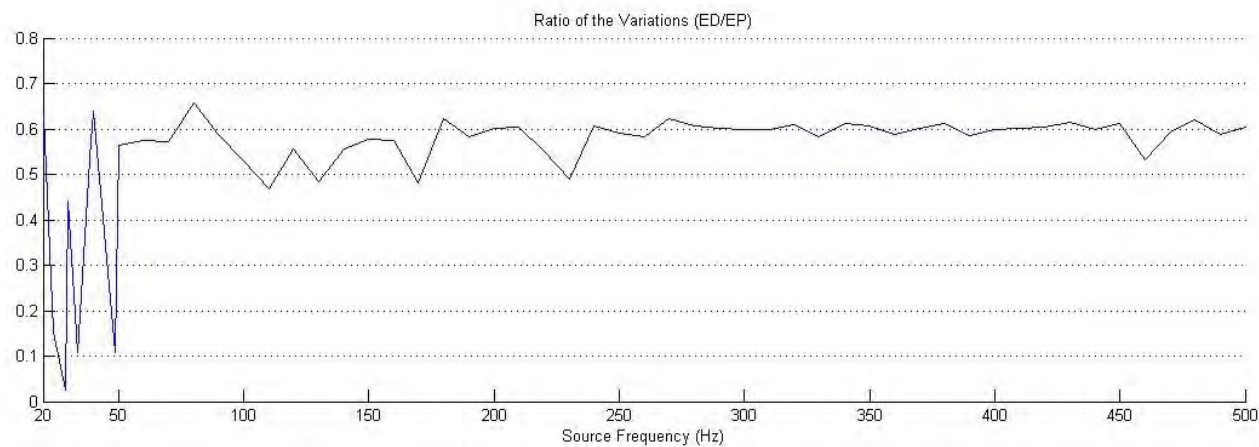
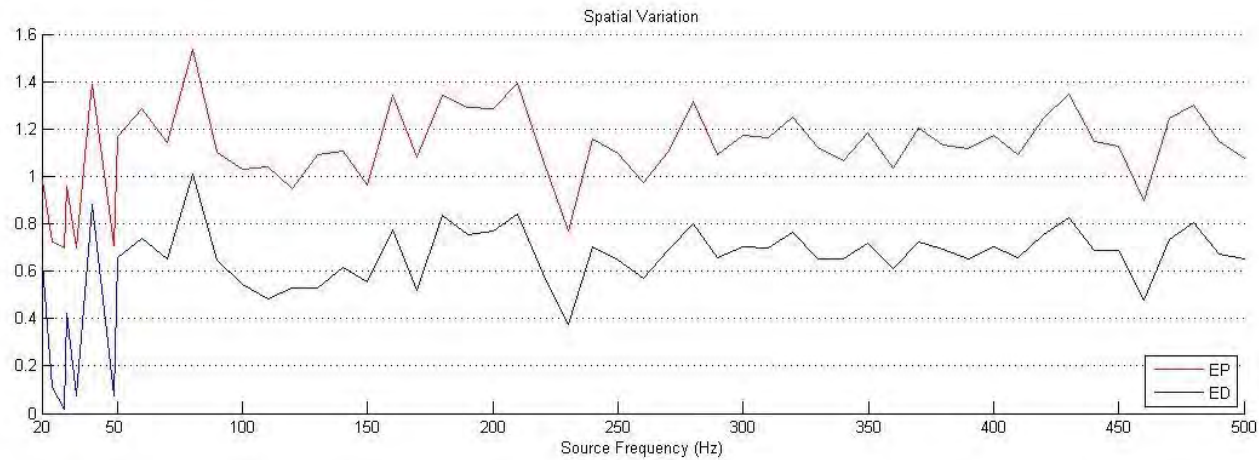


Pressure Response

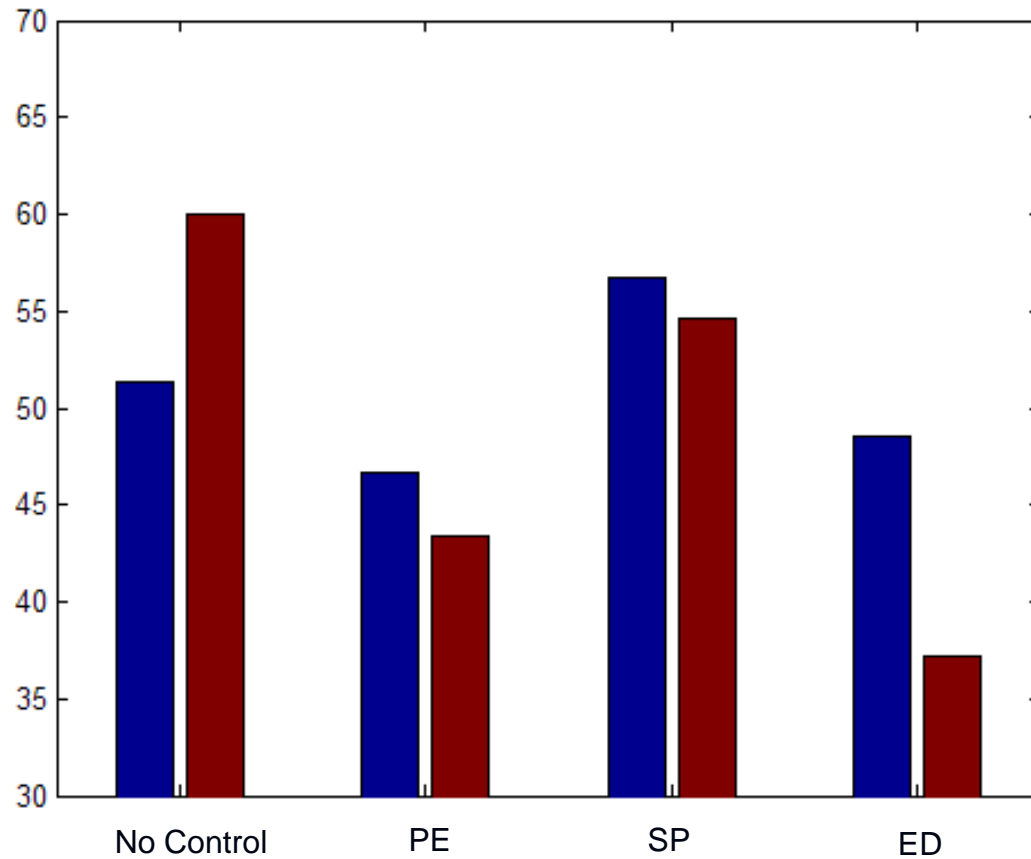


Energy Density Response

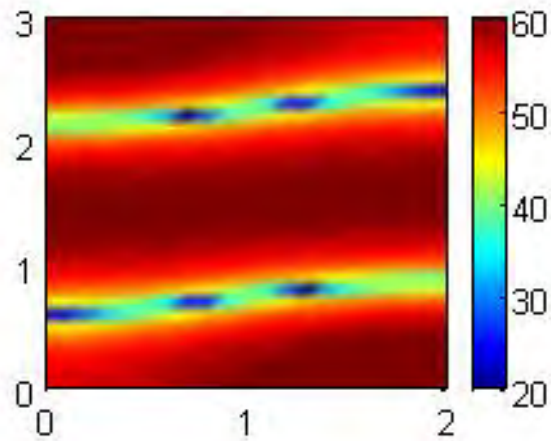
# Spatial Variation of Pressure vs. Energy Density – 3D Field



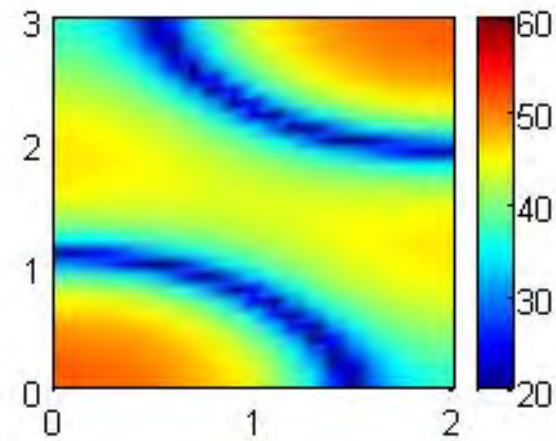
# Example – 2 Mode Case



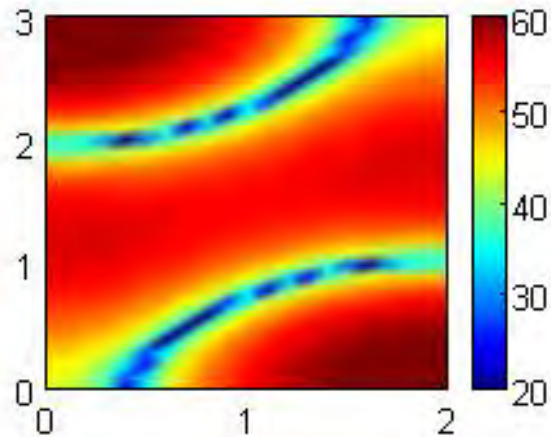
# Example - 2 Mode Case



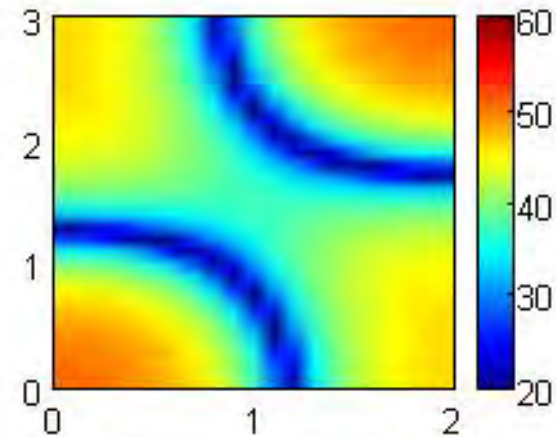
No Control



PE



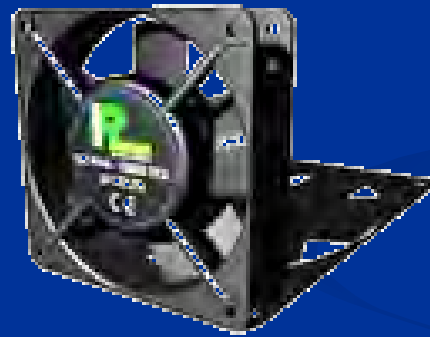
SP



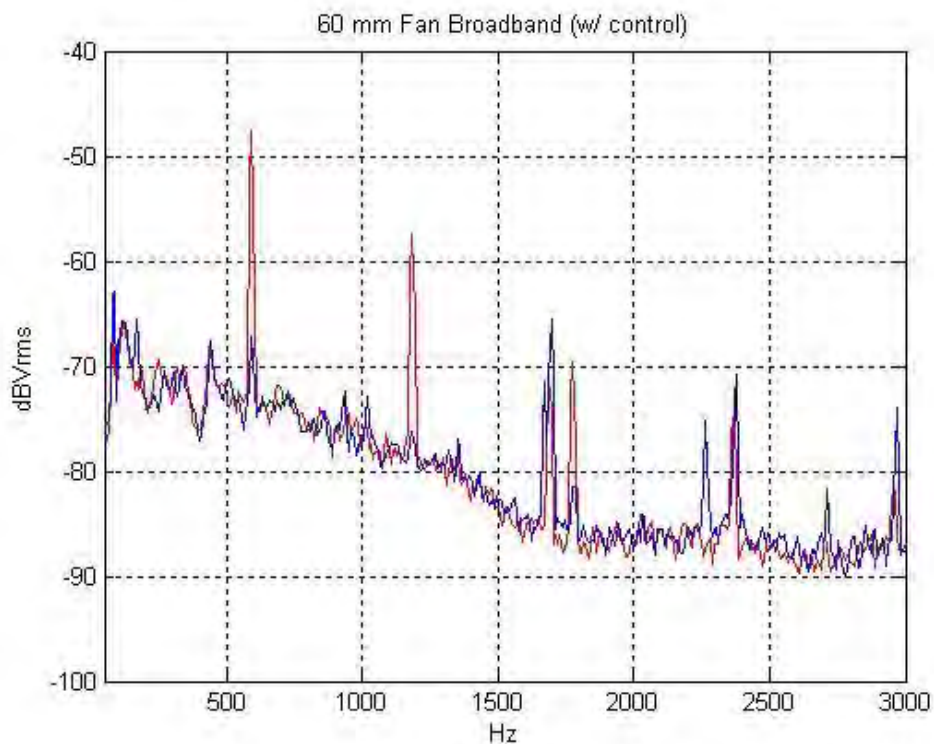
ED

# Application – Axial Fan Noise

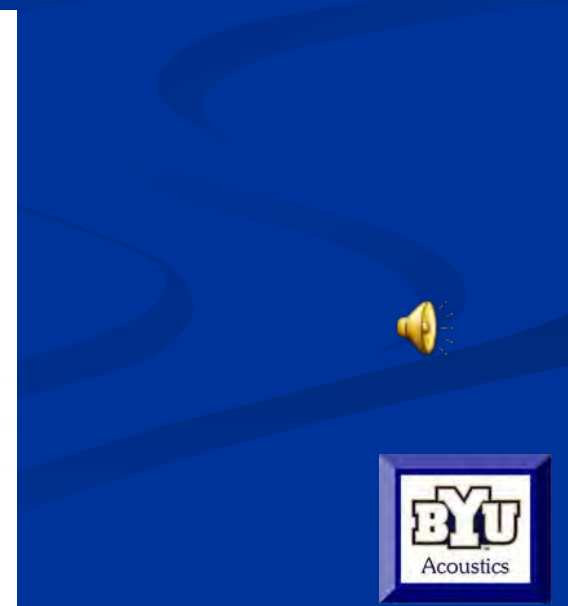
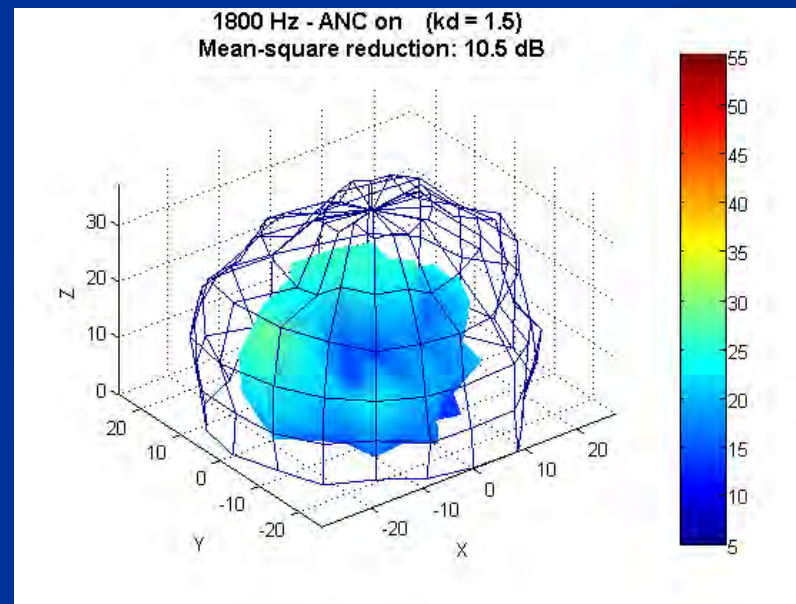
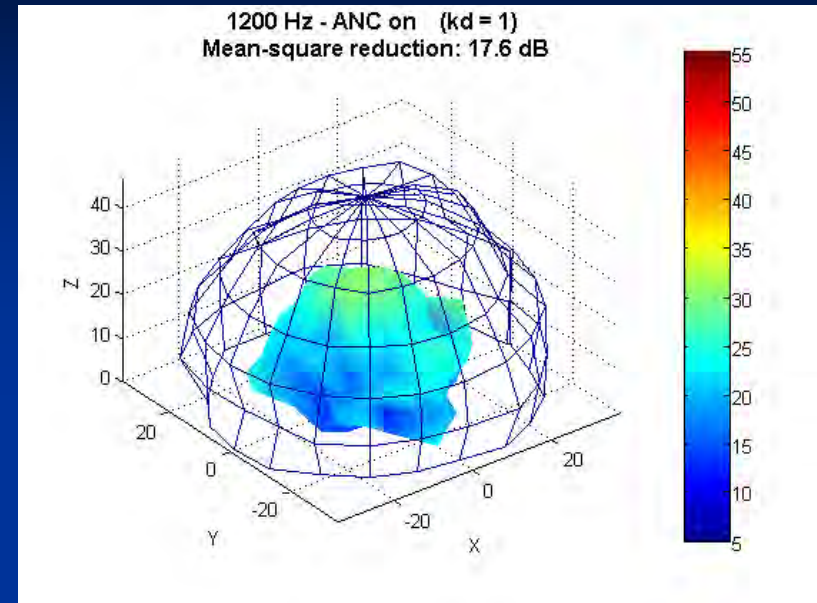
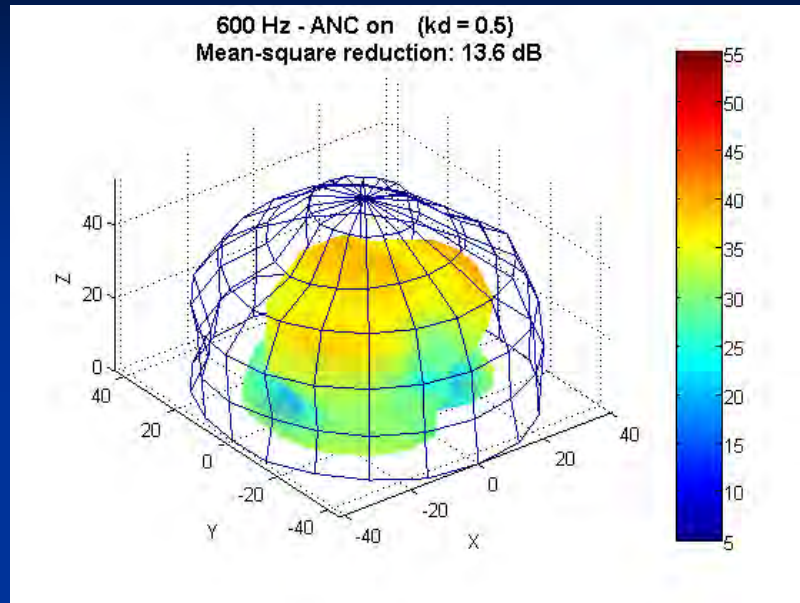
- Seeking global attenuation of radiated fan noise (tonal)
- 60 mm and 80 mm fans



# Control Results – Far Field Mic



# 60 mm Fan Results - Global Response

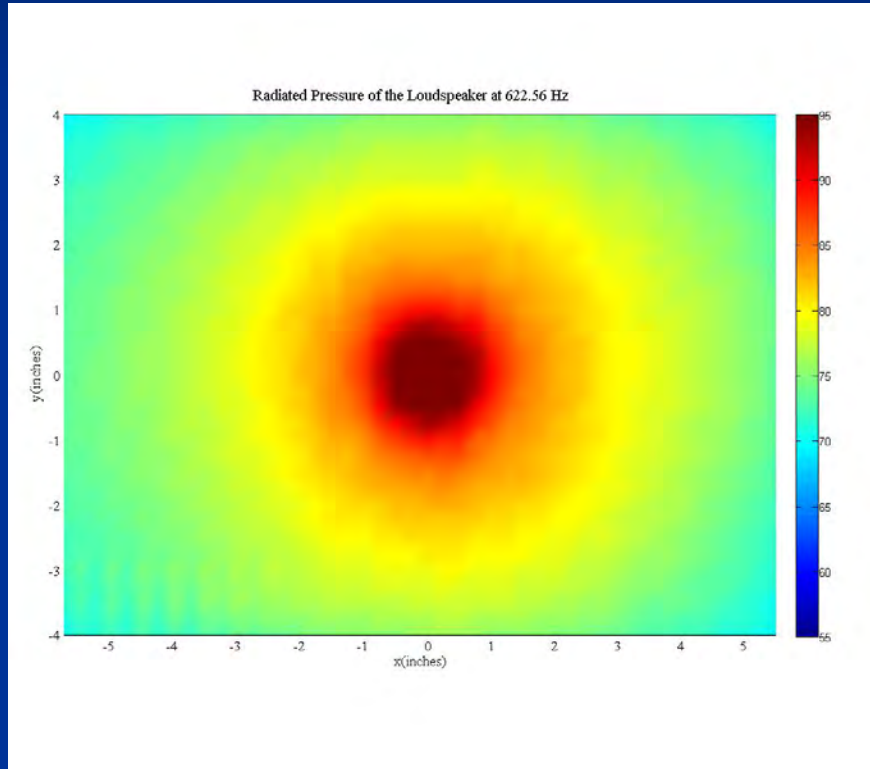


# Fan Results – Global Response

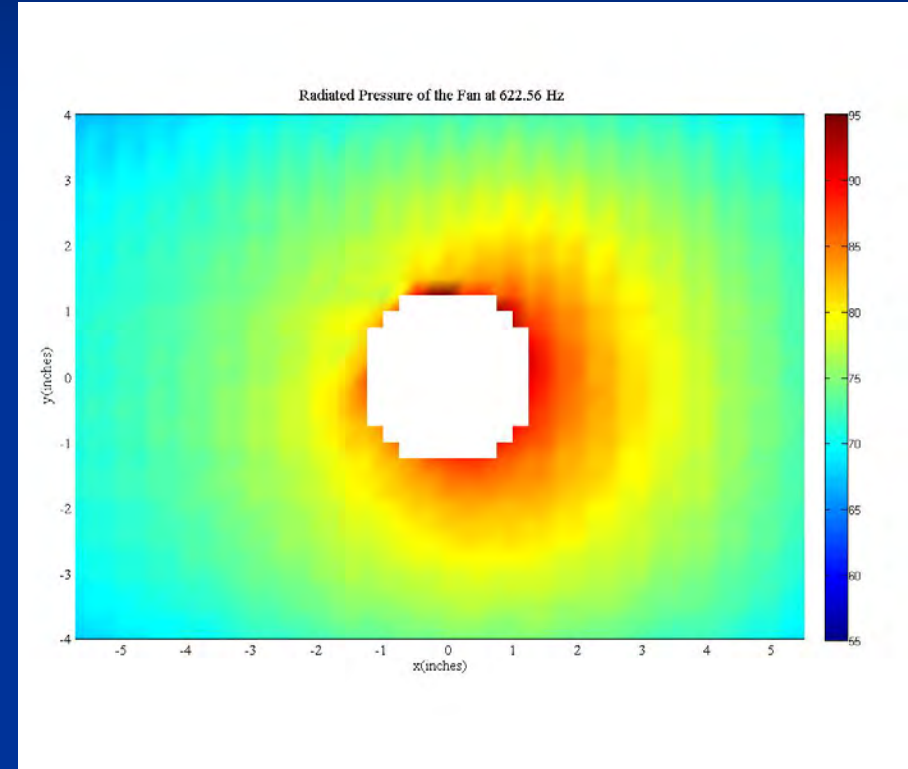
## Global Mean-Square Pressure Reduction

	60 mm		80 mm		Ideal	
	kd	MP R	kd	MPR	kd	MPR
BPF	0.5	13.6	0.4	10.1	0.5	~ 30
2 X BPF	1.0	17.6	0.8	16.1	1.0	22
3 X BPF	1.5	10.5	1.2	12.8	1.5	14

# Near Field Response – No Control



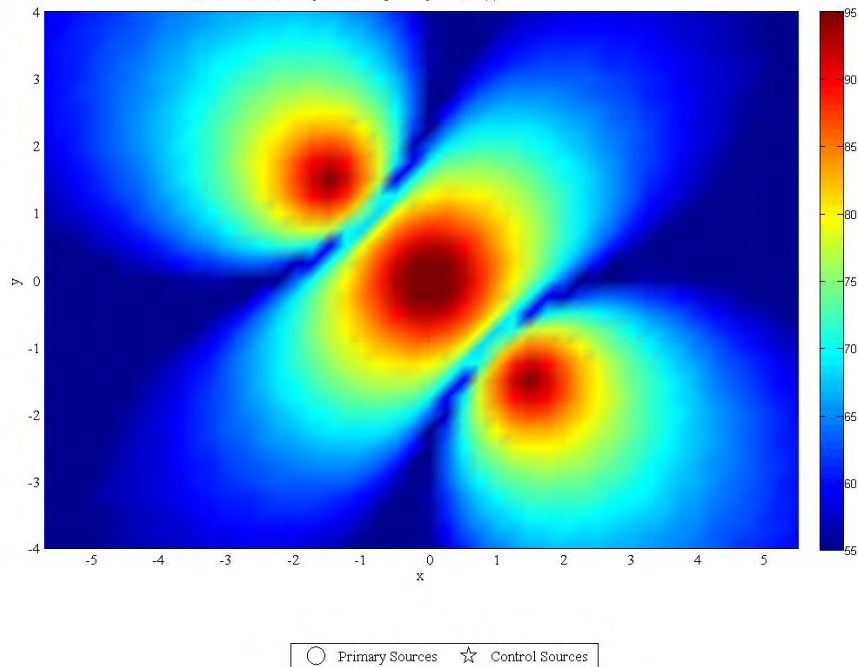
Speaker



60 mm Fan

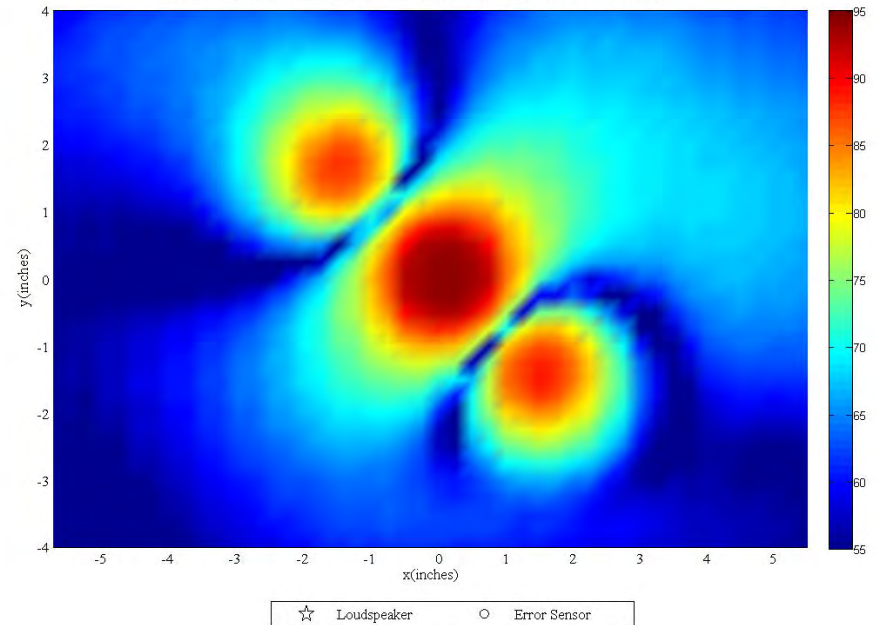
# Near Field Control (Speaker) - 2 Channel Control

Radiated pressure of both primary and secondary source(s) with control at 622 Hz and 0.26 inches away from the primary source(s)



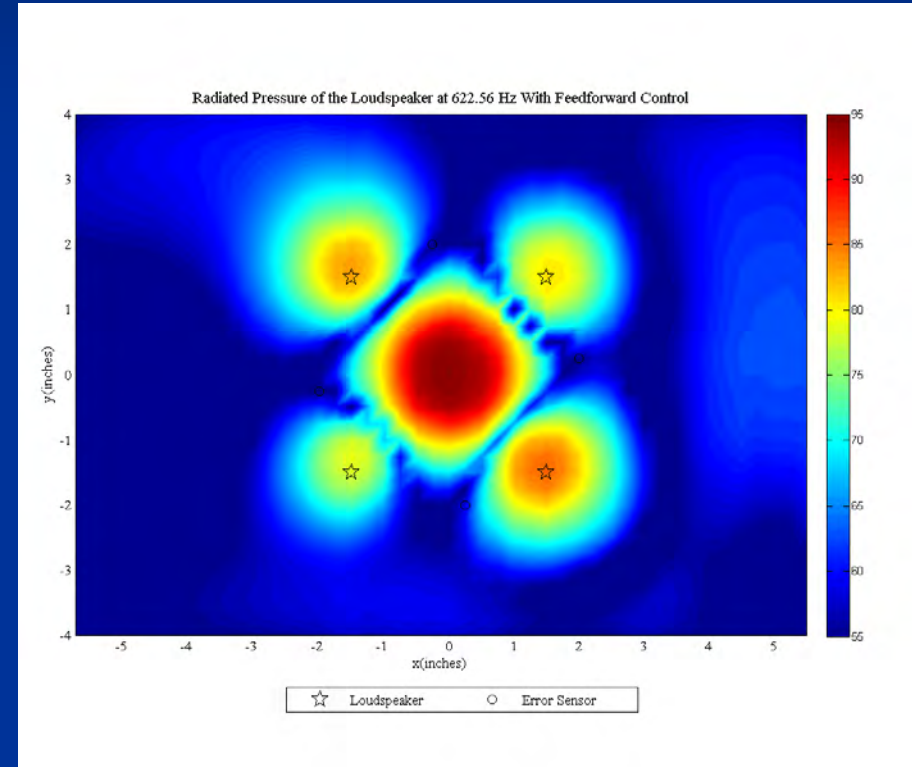
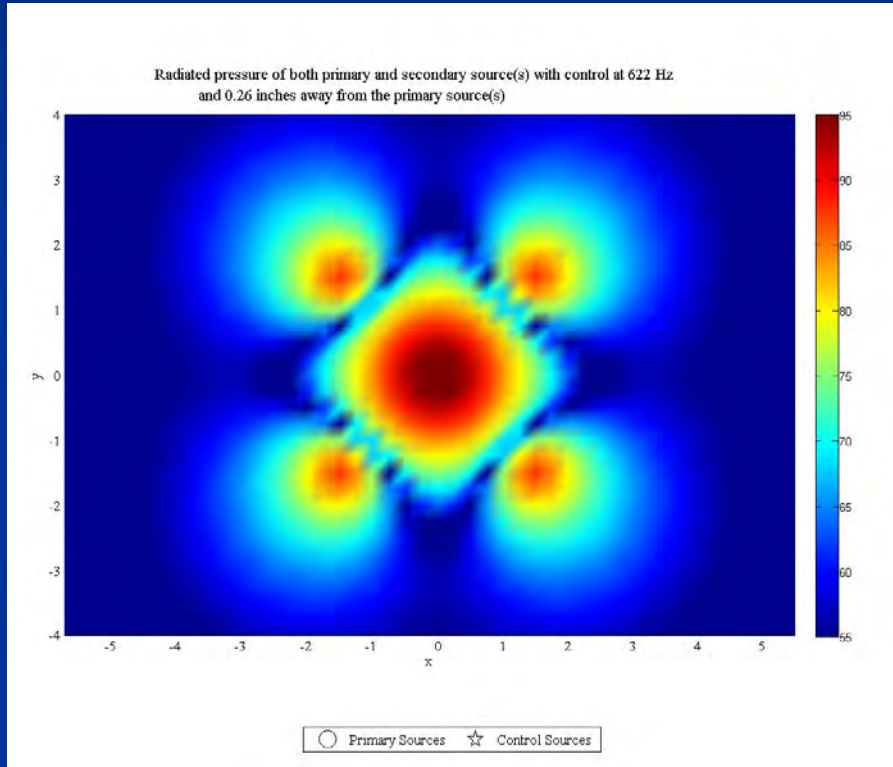
Theoretical Result

Radiated Pressure of the Loudspeaker at 622.56 Hz With Feedforward Control



Measured Result

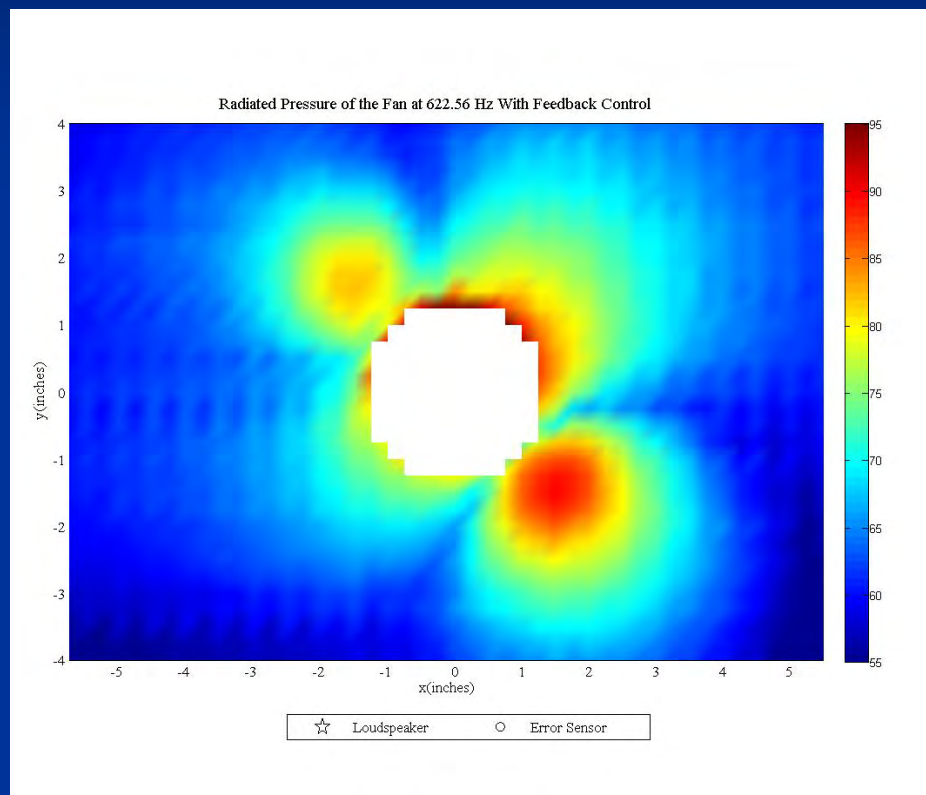
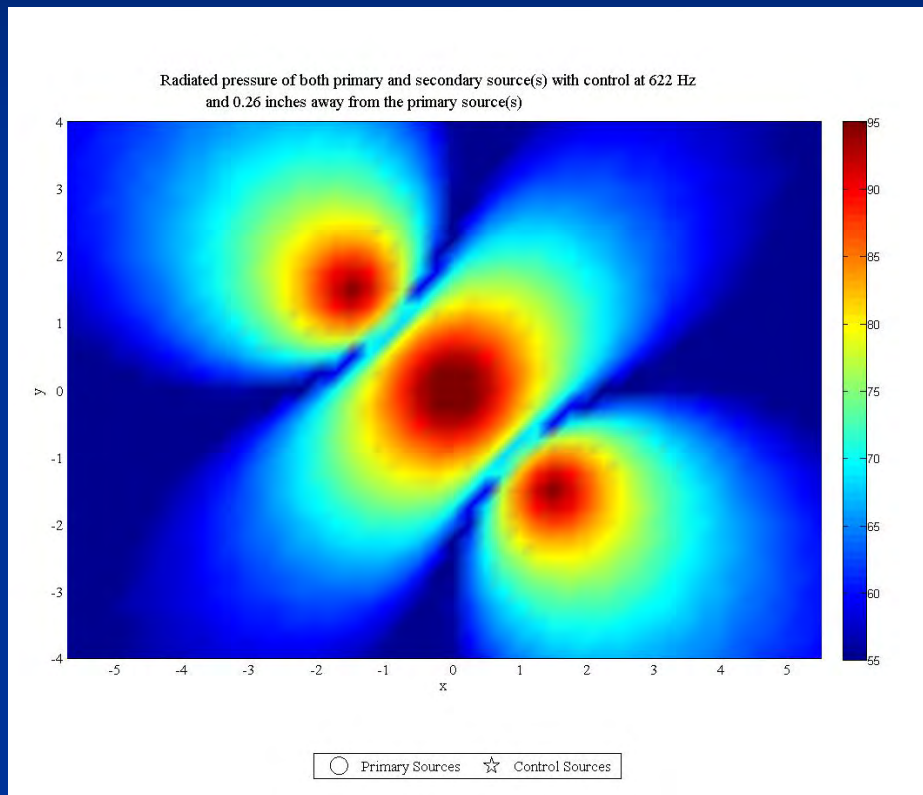
# Near Field Control (Speaker) - 4 Channel Control



Theoretical Result

Measured Result

# Near Field Control (Fan) - 2 Channel Control

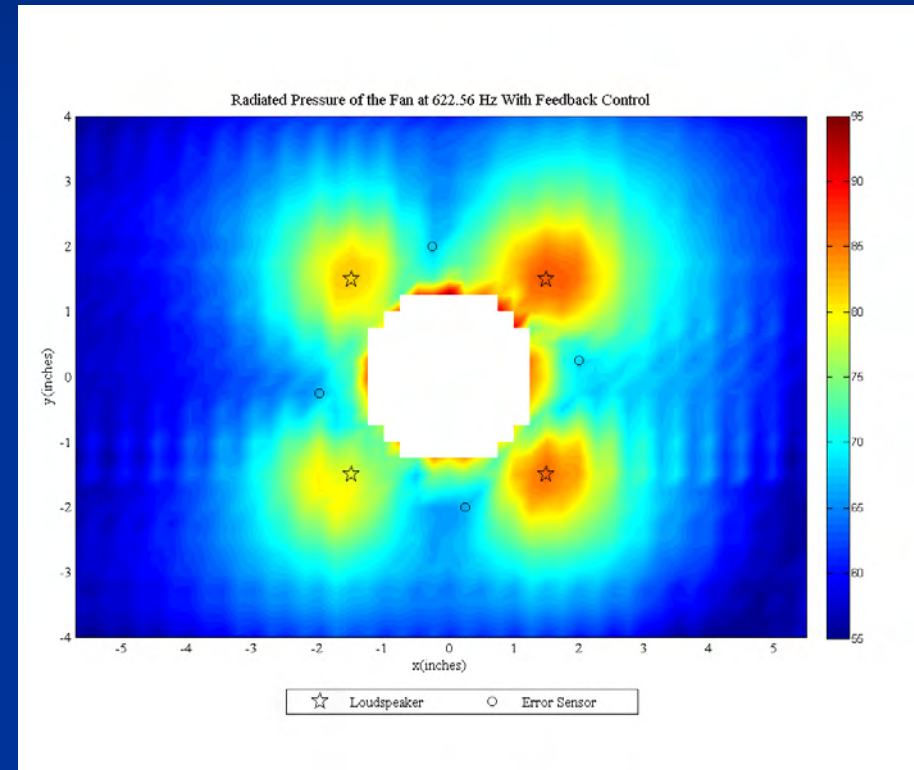
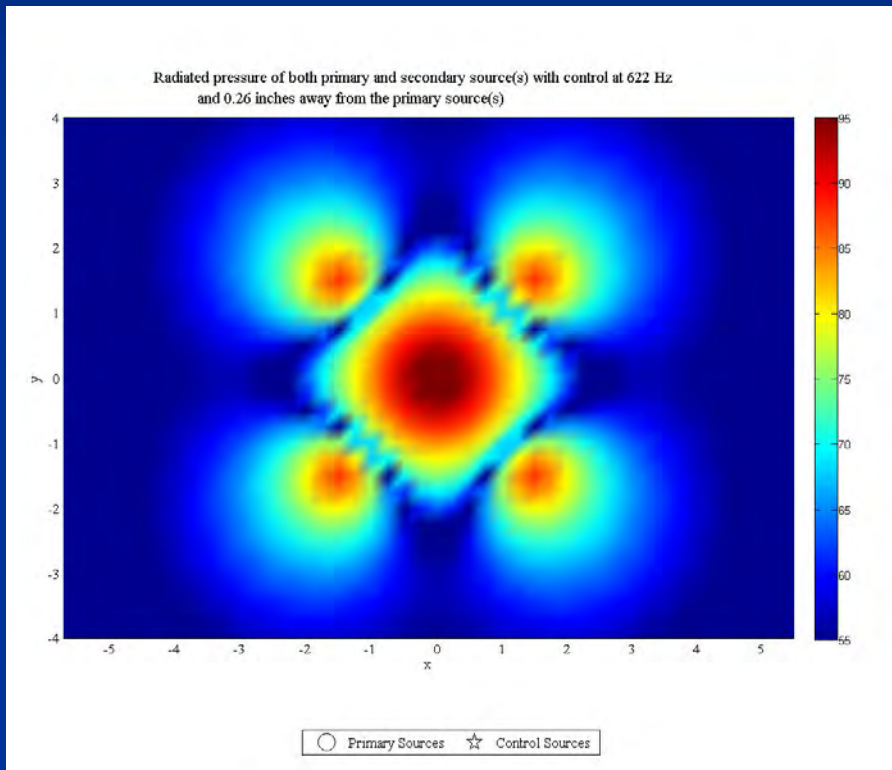


Theoretical Result

Measured Result



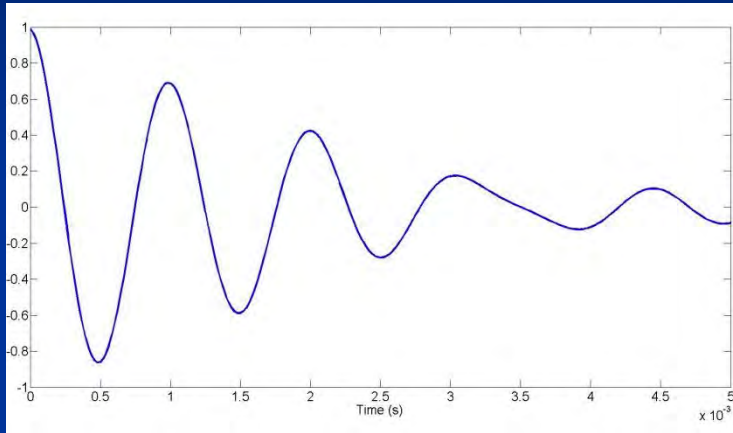
# Near Field Control (Fan) - 4 channel control



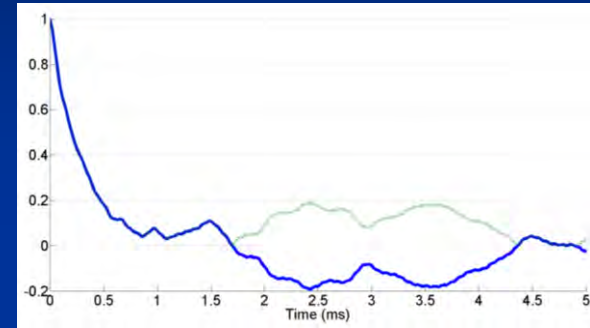
Theoretical Result

Measured Result

# Broadband Control – Theoretical Predictions

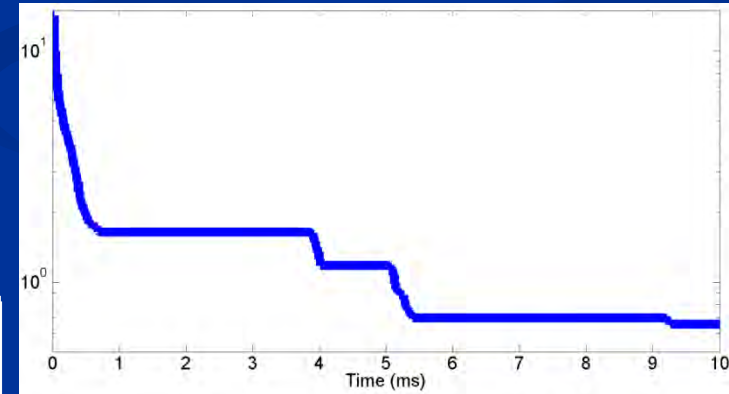


Loudspeaker Autocorrelation

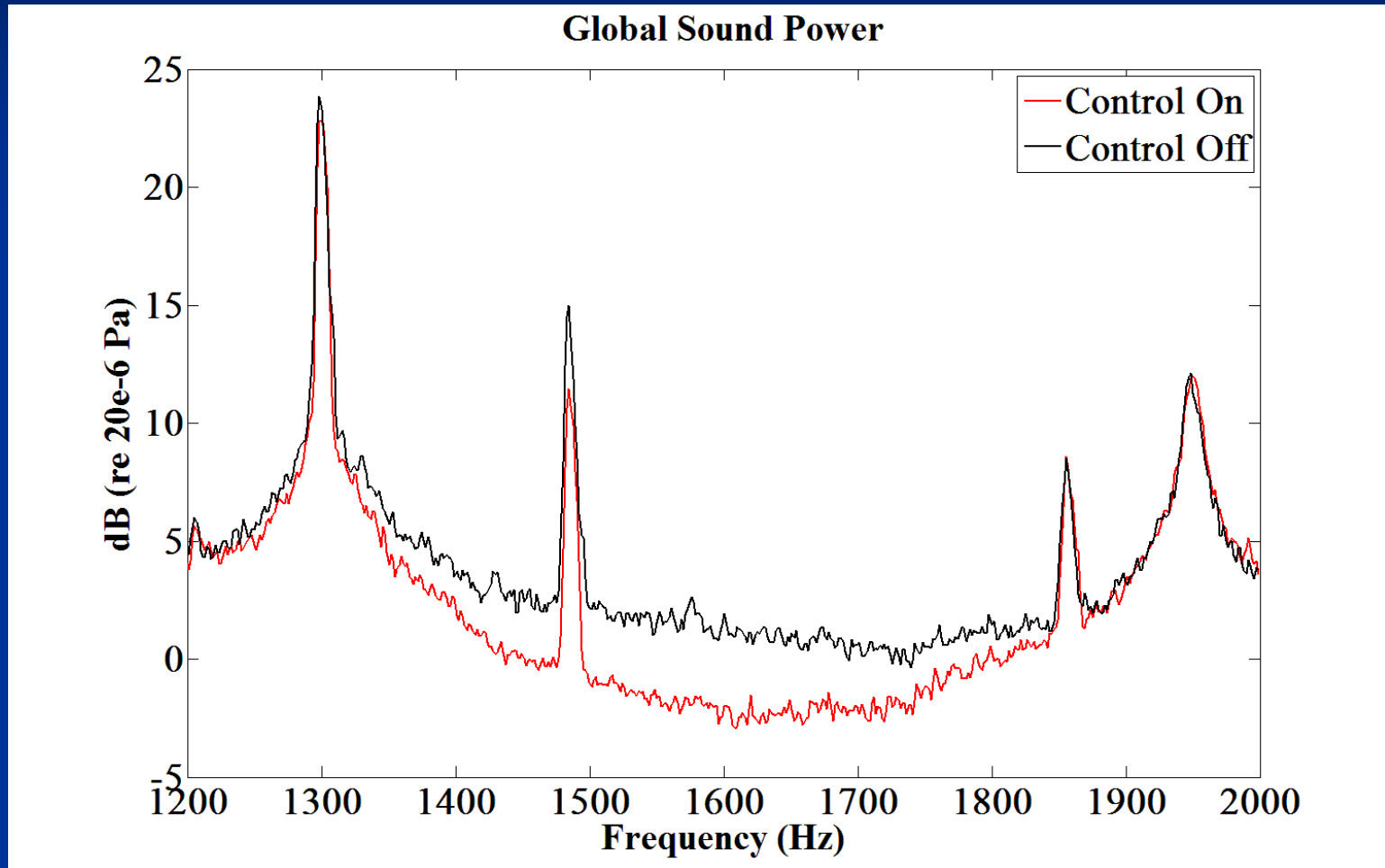


Fan Autocorrelation

$$Attenuation(dB) = -10 \times \log \left( 1 - \frac{E_p}{E_{total}} \right)$$



# Broadband Control - Analog Feedback Control (1.2 dB reduction)



# Application – Wheel Loader Cab Noise

- Interested in tonal noise associated with engine firing frequency



# Filtered-x Algorithm for ED Minimization

- Cost function:

$$J = e = \frac{\hat{p}^2}{2\rho c^2} + \frac{\rho}{2} \hat{v}^2$$

- Gradient of the cost function:

$$\frac{\partial J}{\partial W} = \sum_{m=1}^3 \rho v_m(t) \mathbf{R}_{vm}(t) + \frac{1}{\rho c^2} p(t) \mathbf{R}_p(t)$$

- Update equation:

$$W(t+1) = W(t) - \mu \left( \sum_{m=1}^3 \rho v_m \mathbf{R}_{vm}(t) + \frac{1}{\rho c^2} p(t) \mathbf{R}_p(t) \right)$$

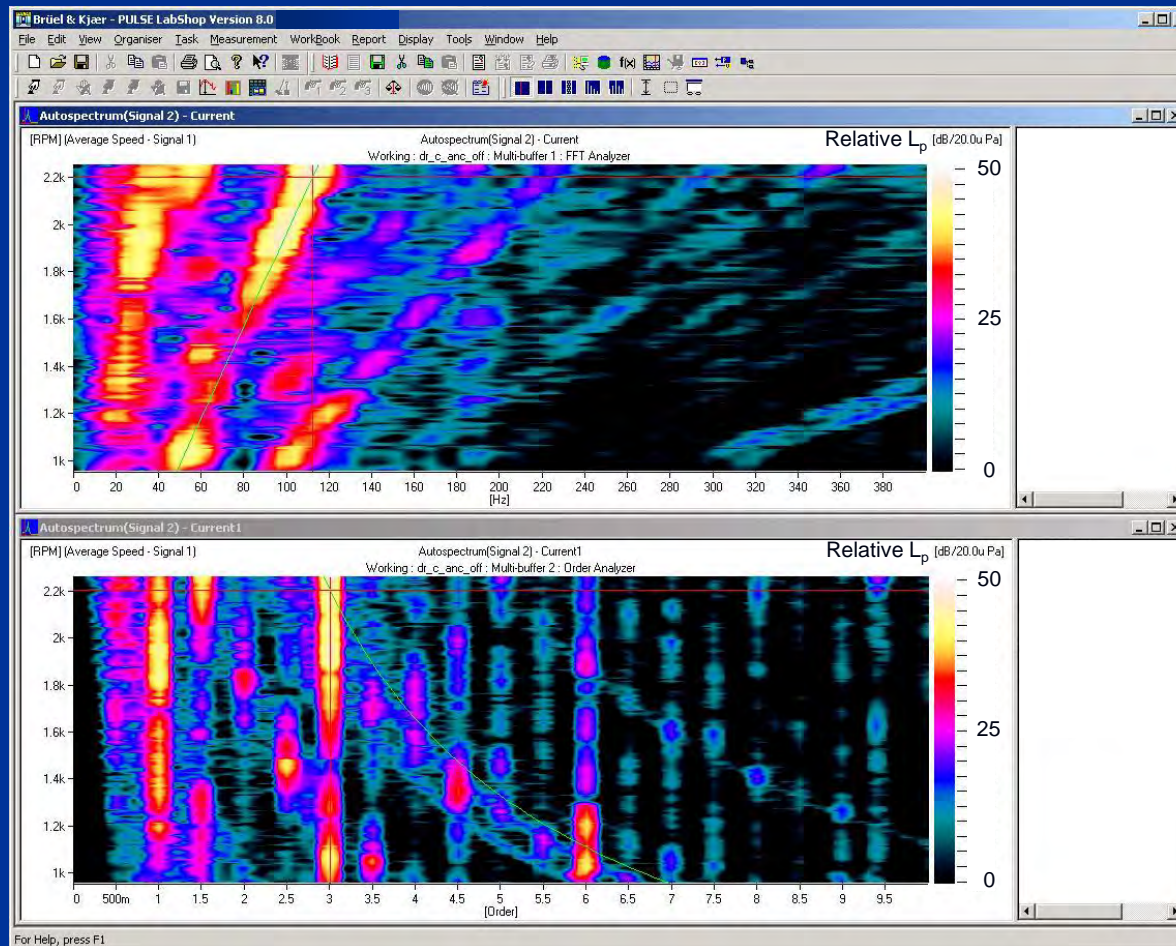
# Hardware Configuration



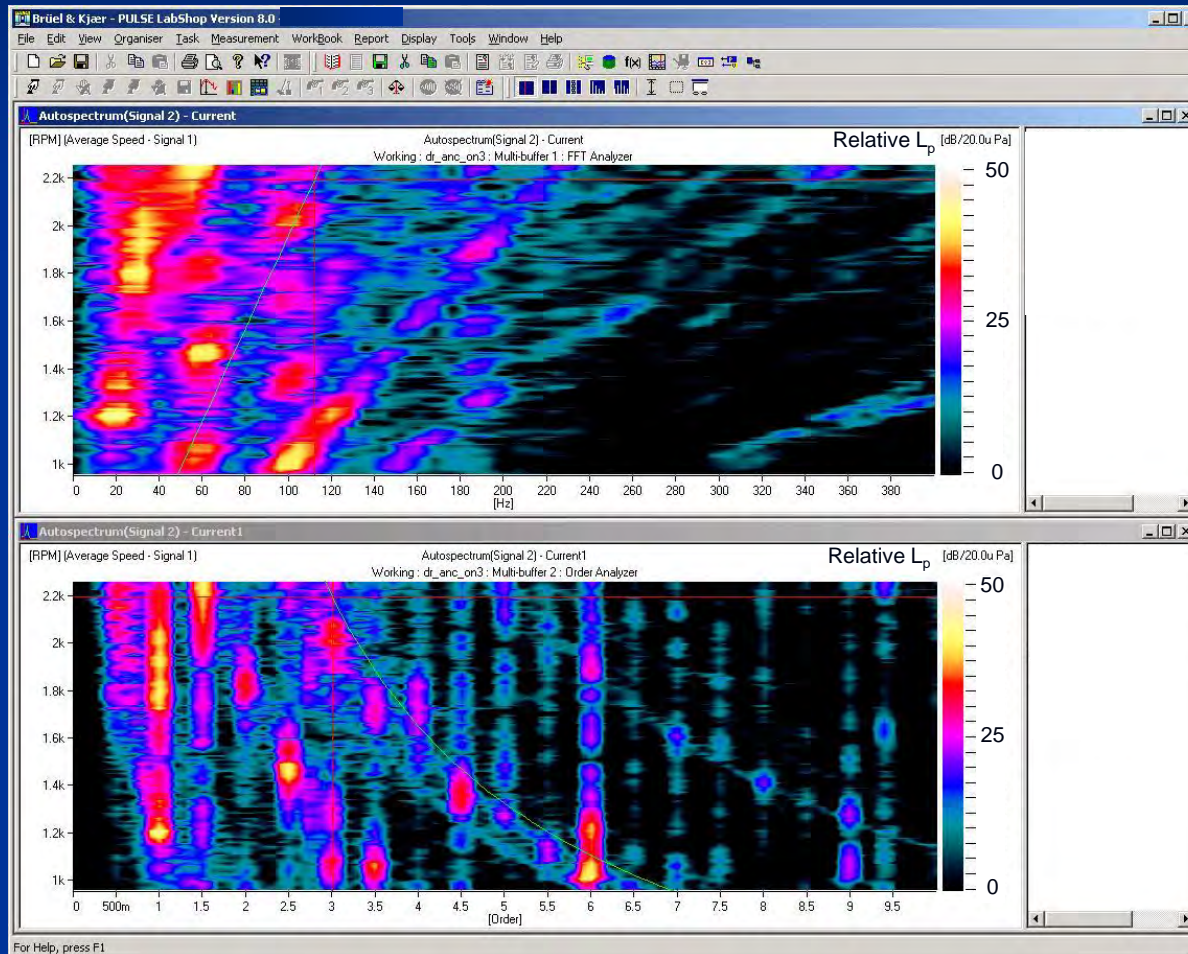
- Sub-woofer used for low frequencies
- Smaller drivers (10 cm) used for higher frequencies
- ED sensor mounted in convenient location

# Cab Response – Engine Sweep

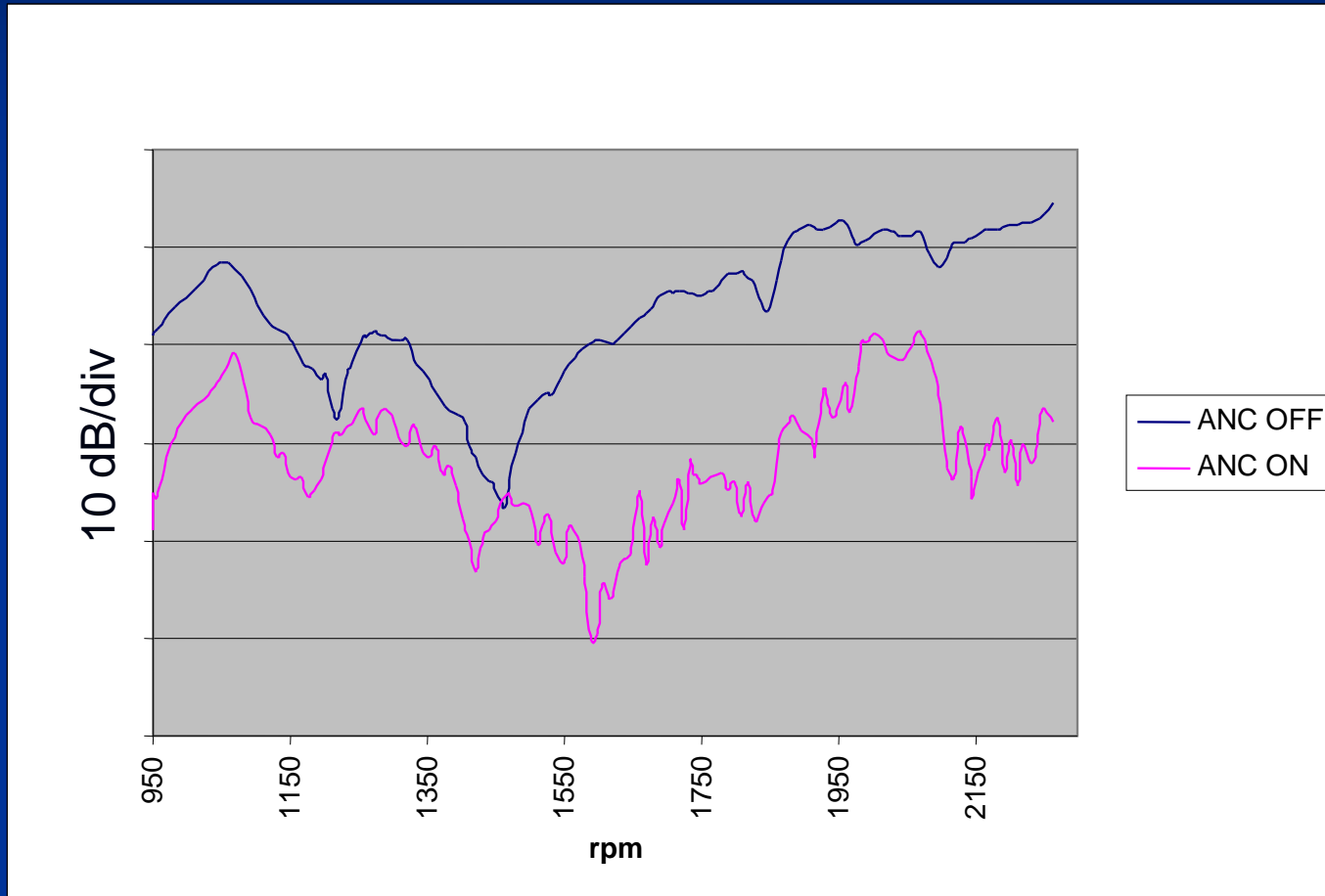
- Results measured at nominal operator head location



# Control Results – Engine Sweep



# Engine Firing Freq - Engine Sweep



# Cab Results – Global Response

- Overall A-weighted attenuation throughout cab



upper plane

front  
↑



lower plane

# Locomotive Noise (NVH Technologies)



Class 66 Locomotive



# Locomotive Noise - Results

- Not only audible perception, but also perceived reduction in whole-body vibration



# Summary

- Initial determination should establish what really needs to be accomplished
- Source configuration (number, type, location) should be established
- Sensor configuration (number, type, location) should be established. Global sensing typically not feasible – what is best compromise?

# Summary – Cont.

- For various applications, sensing techniques have been developed that can provide more global control
- Near field pressure sensors can be used for global control, if care taken as to their placement
- ED sensing is not the “ideal” solution, but does have strong tendency to provide larger region of control – more global in nature